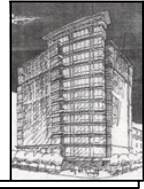


**EXECUTIVE TOWER**  
**NW WASHINGTON, DC**



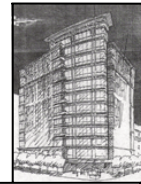
SEAN HOWARD  
STRUCTURAL

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# APPENDIX A

Post Tension  
Hand Checks

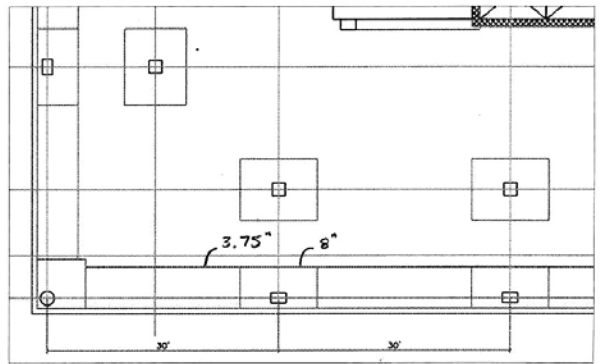
# EXECUTIVE TOWER NW WASHINGTON, DC



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## Material

Normal wt 150 pcf  
 $f'_c = 4000 \text{ psi}$   
 $f'_{ci} = 3000 \text{ psi}$   
 $f_y = 60 \text{ ksi}$   
 $f_{pu} = 270 \text{ ksi}$   
 $f_{se} = 174 \text{ ksi}$   
 $P_{eff} = 26.6 \text{ k/ten}$



## Loading

$LL_0 = 100 \text{ psf}$      $A_T = 255 \text{ ft}^2$   $\therefore$  no LL reduction

Super DL = 20 psf

$$\text{Self wt} = \frac{(6/12)(150)}{255 \text{ ft}^2} + \frac{(4.5' \times 4.5') \left(\frac{8}{12}\right)(150)}{255 \text{ ft}^2} + \frac{(20.5' \times 4') \left(\frac{3.75}{12}\right)(150)}{255 \text{ ft}^2} + \frac{(4 \times 10') \left(\frac{8}{12}\right)(150)}{255}$$

$$= 113.7 \text{ psf}$$

## Slab Section Properties

$$A = 792 \text{ in}^2$$

$$S = \frac{(8.5 \times 12)(6^2)}{6} + \frac{(4 \times 12)(3.75^2)}{6} = 724.5 \text{ in}^3$$

## Parameters

At Jacking

$$f'_{ci} = 3000 \text{ psi}$$

$$\text{comp} = .6 f'_{ci} = 1800 \text{ psi}$$

$$\text{Tension} = 3\sqrt{f'_{ci}} = 164 \text{ psi}$$

At service Loads

$$f'_c = 4000 \text{ psi}$$

$$\text{comp} = .45 f'_c = 1800 \text{ psi}$$

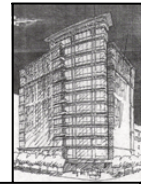
$$\text{tension} = 6\sqrt{f'_c} = 380 \text{ psi}$$

Ave Precamp

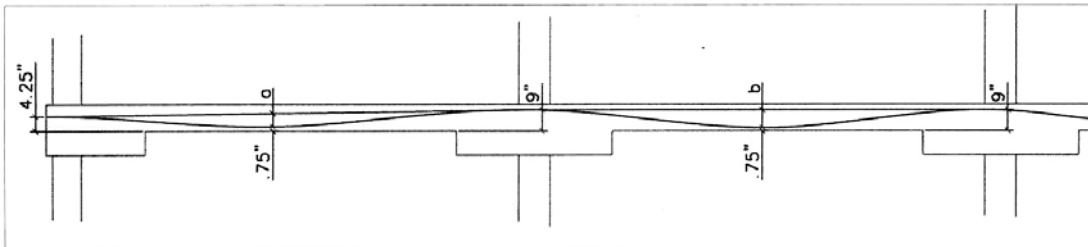
$$P/A = 125 \text{ psi min}$$

1/4

# EXECUTIVE TOWER NW WASHINGTON, DC



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$$a = \left(\frac{4.25 + 9}{2}\right) - .75 = 5.875''$$

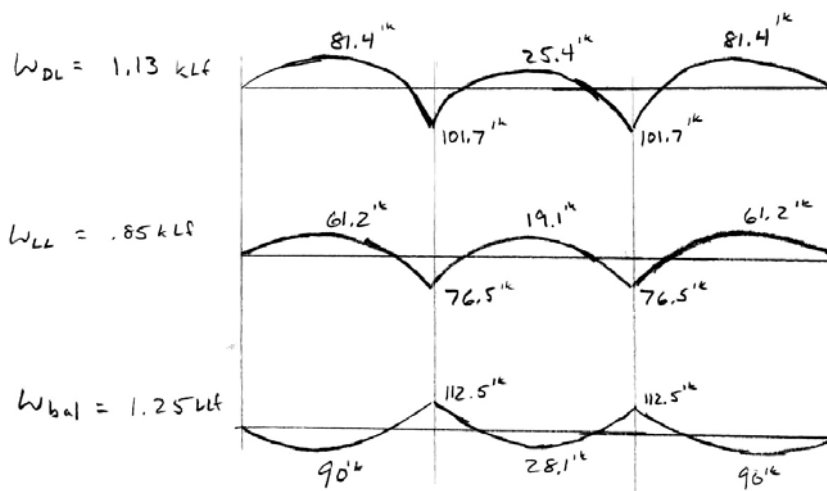
$$b = 9 - .75 = 8.25''$$

$$P = (9 \text{ tendon})(26.6^k) = 239^k$$

$$P/A = \frac{239(1000)}{792 \text{ in}^2} = 302 \text{ psi} > 125 \text{ psi}$$

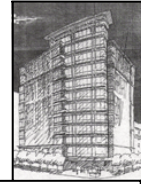
$$w_b = \frac{(239^k)(8)(5.87/12'')}{30^2} = 1.04 \text{ klf}$$

$$= \frac{239(8)(8.25/12'')}{30^2} = 1.46 \text{ klf}$$



2/4

# EXECUTIVE TOWER NW WASHINGTON, DC



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## After jacking

Interior span

midspan stress

$$f_{top} = \frac{(+25.4 + 28.1)(12)(1000)}{724 \text{ in}^3} - 302 \text{ psi} = -346 \text{ psi} < -1800 \text{ psi ok}$$

$$f_{bot} = \frac{(-25.4 + 28.1)(12)(1000)}{724 \text{ in}^3} - 302 \text{ psi} = -257 \text{ psi} < -1800 \text{ psi ok}$$

end stress

$$f_{top} = \frac{(112.5 - 101.7)(12)(1000)}{724} - 302 = -123 \text{ psi} < -1800 \text{ psi ok}$$

$$f_{bot} = \frac{(-112.5 + 101.7)(12)(1000)}{724} - 302 = -481 \text{ psi} < 1800 \text{ psi ok}$$

Exterior spans

midspan stress

$$f_{top} = \frac{(81.4 - 90)(12)(1000)}{724} - 302 = -444 \text{ psi} < 1800 \text{ psi ok}$$

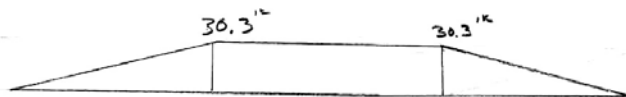
$$f_{bot} = \frac{(-81.4 + 90)(12)(1000)}{724} - 302 = -159.4 \text{ psi} < 1800 \text{ psi ok}$$

## Ultimate Strength

$$M_i = P \cdot e$$

$$= 239 \cdot (9 - 4.875) / 12 = 82.1 \text{ k}$$

$$M_{sec} = M_{bal} - M_i = 112.5 - 82.1 = 30.3 \text{ k}$$



at midspan

$$M_u = 1.2(81.4) + 1.6(61.2) + 1.0\left(\frac{30.3}{2}\right) = 210.7 \text{ k}$$

at support

$$M_u = 1.2(-101.7) + 1.6(-76.5) + 1.0(30.3) = -214.1 \text{ k}$$

3/4

# EXECUTIVE TOWER

## NW WASHINGTON, DC



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### Reinforcement

Bottom & Top bar

		CS	ft-k	ft	ft-k/ft	#4	check?
mid	210.7	.60	126.4	4'	31.6	95.6	good
		.40	84.3	4.5'	18.7	55.1	good
support	214.1	.75	160.6	4'	40.15	95.6	good
		.25	53.5	4.5'	11.8	55.1	good

#4 @ 12"

C.S

$$A_s = .2 \text{ in}^2/\text{ft}$$

$$d = 9"$$

$$a = \frac{.20(60)}{.85(4)(12)} = .29 \text{ in}$$

$$\phi M_n = \phi (.2)(60)(9.0 - \frac{.29}{2})$$

$$\phi M_n = 95.6 \text{ k}$$

M.S

$$A_s = .2 \text{ in}^2/\text{ft}$$

$$d = 5.25$$

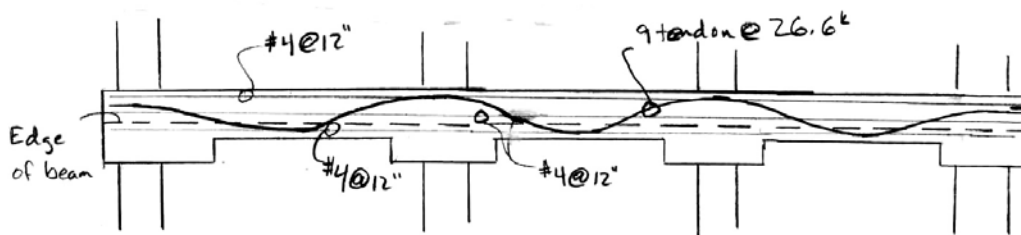
$$a = \frac{.20(60)}{.85(4)(12)} = .29 \text{ in}$$

$$\phi M_n = \phi (.2)(60)(5.25 - \frac{.29}{2})$$

$$\phi M_n = 55.1 \text{ k}$$

USE #4 @ 12" o.c.

+ 9 tendons AT



4/4

# EXECUTIVE TOWER NW WASHINGTON, DC



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Assuming curve to act  
as a straight system

## Materials

Normal wt 150 pcf

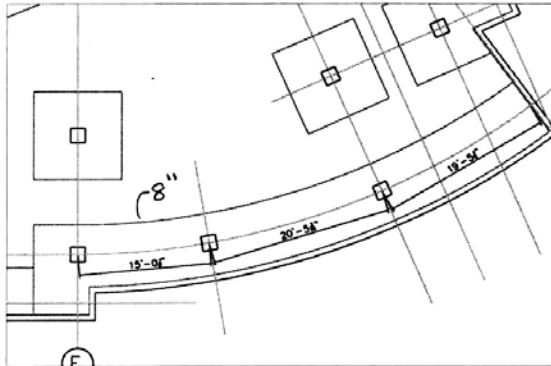
$$f'_c = 4000 \text{ psi}$$

$$f'_{ci} = 3000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$f_{pu} = 270 \text{ ksi}$$

$$P_{eff} = 26.6 \text{ k/ton}$$



## Loading

$$LL_o = 100 \text{ psf}$$

$$\text{Sup DL} = 20 \text{ psf}$$

$$\text{Self wt} = (6/12)(150) + \frac{(7 \times 55)(\frac{8}{12})(150)}{632.5 \text{ ft}^2} = 135.8 \text{ psf}$$

$$A_T = 235.7 \text{ ft (for largest bay } \therefore \text{ no LL Reduction)}$$

$$\text{Ave width} = 11.5'$$

## Slab Properties

$$A = 1500 \text{ in}^2$$

$$S = \frac{(11.5 \times 12)(6)^2}{6} + \frac{(7 \times 12)(8)^2}{6} = 1724 \text{ in}^3$$

## Parameters

At jacking

$$f'_{ci} = 3000 \text{ psi}$$

$$\text{comp} = .6 f'_{ci} = 1800 \text{ psi}$$

$$\text{Tension} = 3\sqrt{f'_{ci}} = 164 \text{ psi}$$

At service loads

$$f'_c = 4000 \text{ psi}$$

$$\text{comp} = .45 f'_c = 1800 \text{ psi}$$

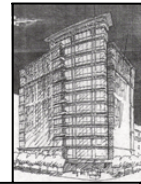
$$\text{tension} = 6\sqrt{f'_c} = 380 \text{ psi}$$

Ave Precomp

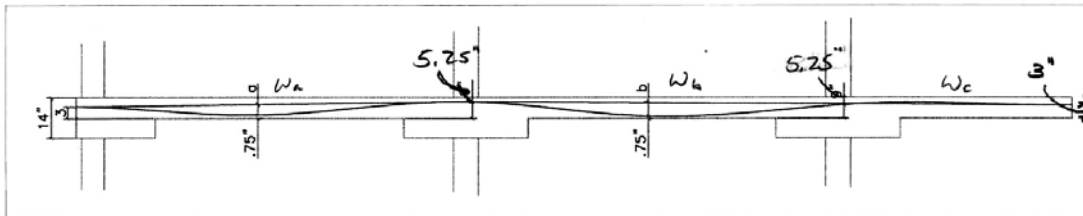
$$P/A = 125 \text{ psi min}$$

1/4

# EXECUTIVE TOWER NW WASHINGTON, DC



SEAN HOWARD  
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$$a = \frac{(5.25 + 3)}{2} - .75 = 3.37''$$

$$b = 4.5''$$

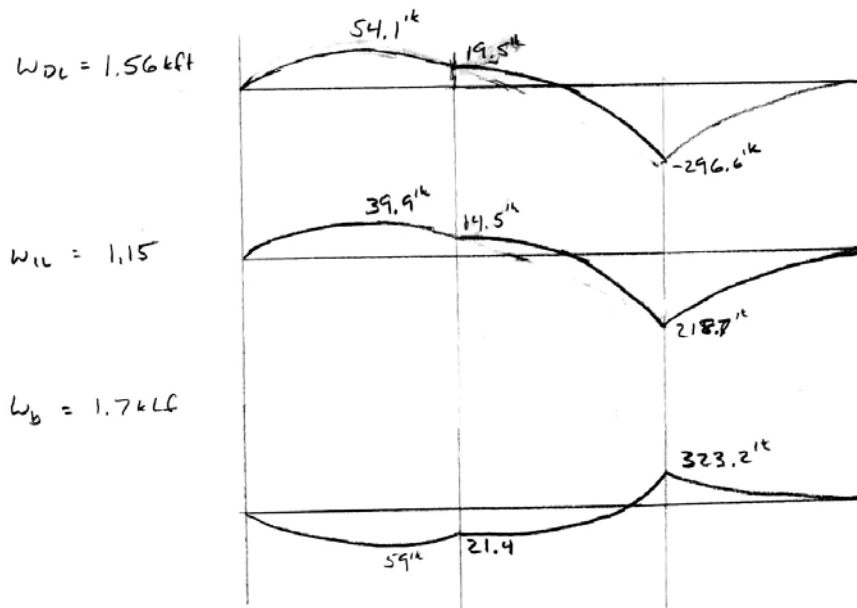
$$P = (\text{tendon}) (26.6) = 239^k$$

$$P/A = \frac{239(1000)}{1500 \text{ in}^2} = 159.3 \text{ psi} > 125 \text{ ok}$$

$$W_a = \frac{(239)(8)(3.37''/12)}{15^k} = 2.39 \text{ klf}$$

$$W_b = 1.7 \text{ klf}$$

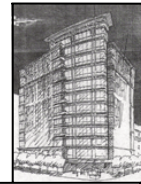
$$W_c = .94 \text{ klf}$$



2/4

# EXECUTIVE TOWER

## NW WASHINGTON, DC



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STRUCTURAL

After jacking

"15' foot" span

midspan stress

$$f_{top} = \frac{(54.1 - 59)(12)(1000)}{1724 \text{ in}^3} - 159.3 \text{ psi} = -193.1 \text{ psi} < 1800 \text{ psi } \underline{ok}$$

$$f_{bot} = \frac{(-54.1 + 59)(12)(1000)}{1724} - 159.3 = -125.2 \text{ psi} < 1800 \text{ psi } \underline{ok}$$

support

$$f_{top} = \left( \frac{19.5 - 21.4}{1724} \right) (12)(1000) - 159.3 = -172.5 \text{ psi} < 1800 \text{ psi } \underline{ok}$$

$$f_{bot} = +13.2 - 159.3 = -146.1 \text{ psi} < 1800 \text{ psi } \underline{ok}$$

"20.5' / 19.5" span

support

$$f_{top} = \frac{(-296.6 + 323.2 \text{ k})(12)(1000)}{1724} - 159.3 = +25.8 \text{ psi} < 164 \text{ psi tension } \underline{ok}$$

$$f_{bot} = -25.8 - 159.3 = -185.1 \text{ psi} < 1800 \text{ psi } \underline{ok}$$

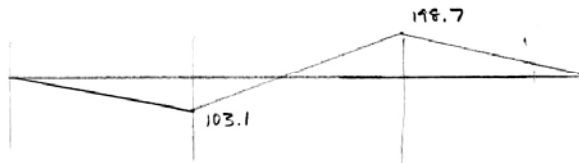
Ultimate Strength

$$M_1 = P \cdot e$$

$$= 239(6.25)/12 = 124.5 \text{ k}$$

$$M_{sec} = M_{bal} - M_1 = -103.1 \text{ k}$$

$$= +198.7 \text{ k}$$



"15' foot"

$$Mid M_u = 1.2(54.1 \text{ k}) + 1.6(39.9 \text{ k}) + 1.0\left(\frac{103.1}{2}\right) = 77.2 \text{ k}$$

$$sup M_u = 1.2(19.5) + 1.6(14.5) + 1.0(-103.1) = -56.5 \text{ k}$$

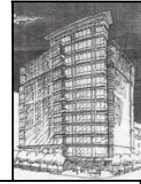
"20.5' / 19.5"

$$sup M_u = 1.2(-296.6) + 1.6(-218.7) + 1.0(198.7) = -507.14$$

3/4



# EXECUTIVE TOWER NW WASHINGTON, DC



SEAN HOWARD  
STRUCTURAL

## Reinforcement

		ft-k	ft	ft-k/ft	# 4	check	
77.2	CS .60	46.3	7'	6.6	11.8	ok	ft-k/ft ①
	MS .40	30.8	4.5'	6.8	4.6	2.2	
-56.5	CS .75	-42.4	7'	-6.0	-11.8	ok	ft-k/ft ②
	MS .25	-14.1	4.5'	-3.1	-4.6	ok	
-507.1	CS .75	-380.3	7'	-54.3	-11.8	42.5	ft-k/ft ③
	MS .25	-126.8	4.5'	-28.17	-4.6	24.1	ft-k/ft ③

#4 @ 12" o.c.

C.S.  
 $A_s = .20 \text{ in}^2/\text{ft}$   
 $d = 13.25$

$$a = \frac{.20(60)}{.85(4)(12)} = .29 \text{ in}$$

$$\phi M_n = \phi(.2)(60)(13.25 - \frac{.29}{2})/12'$$

$$\phi M_n = 11.8 \text{ k/ft}$$

M.S.

$$A_s = .2 \text{ in}^2/\text{ft}$$

$$d = 5.25$$

$$a = .29 \text{ in}$$

$$\phi M_n = \phi(.2)(60)(5.25 - \frac{.29}{2})/12'$$

$$\phi M_n = 4.6 \text{ k/ft}$$

① try 1#5

$$M_n = (2.2 \text{ k/ft})(4.5') = 9.9 \text{ k}$$

$$a = \frac{.31(60)}{.85(4)(4.5)(12)} = .10 \text{ in}$$

$$\phi M_n = \phi(.31)(60)(5.25 - \frac{.10}{2})/12'$$

$$\phi M_n = 7.25 < 9.9 \text{ k USE } 2\#5$$

② try 2#8

$$M_n = (42.5)(7') = 297.5 \text{ k}$$

$$a = \frac{6.28(60)}{.85(4)(7)(12)} = 1.32 \text{ in}$$

$$\phi M_n = \phi(6.28)(60)(13.0 - \frac{1.32}{2})/12'$$

$$\phi M_n = 348.7 > 297.5 \text{ k ok}$$

③ try 2#8

$$M_n = (24.1)(4.5) = 108.4 \text{ k}$$

$$a = \frac{6.28(60)}{.85(4)(4.5)(12)} = 2.05 \text{ in}$$

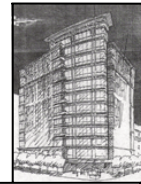
$$\phi M_n = \phi(6.28)(60)(5 - \frac{2.05}{2})/12'$$

$$\phi M_n = 113.0 \text{ k} > 108.4 \text{ k ok}$$

4/4

# EXECUTIVE TOWER

## NW WASHINGTON, DC



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STRUCTURAL

### Materials

Normal wt conc 150pcf

$$f'_c = 4000 \text{ psi}$$

$$f'_{ci} = 3000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$f_{pu} = 270 \text{ ksi}$$

$$P_{efc} = 26.6 \text{ k/ten}$$

### Loading

$$LL_o = 100 \text{ psf}$$

$$A_T = 2341 \text{ ft}^2$$

$$LL = LL_o \left( .25 + \frac{15}{\sqrt{A_T}} \right)$$

$$= 100 \left( .25 + \frac{15}{\sqrt{2341}} \right) = 56.0 \text{ psf}$$

$$\text{Sup Dead} = 20 \text{ psf}$$

### Slab Properties

$$A = 1260 \text{ ft}^2$$

$$S = 1260 \text{ ft}^2$$

### Parameters

At jacking

$$f'_{ci} = 3000 \text{ psi}$$

$$\text{comp} = .6 f'_{ci} = 1800 \text{ psi}$$

$$\text{tension} = 3 \sqrt{f'_{ci}} = 164 \text{ psi}$$

At service loads

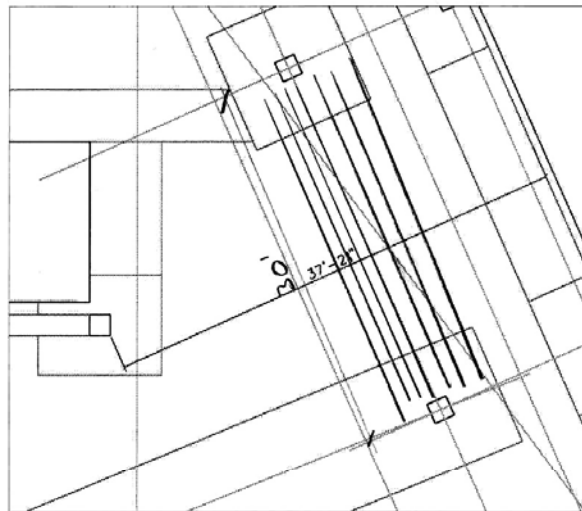
$$f'_c = 4000 \text{ psi}$$

$$\text{comp} = .45 f'_c = 1800 \text{ psi}$$

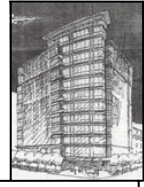
$$\text{tension} = 6 \sqrt{f'_c} = 380 \text{ psi}$$

Ave Precomp

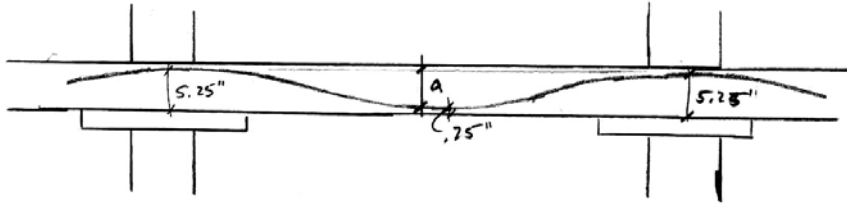
$$P/A = 125 \text{ psi min}$$



# EXECUTIVE TOWER NW WASHINGTON, DC



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$$a = 5.25'' - .75 = 4.5''$$

$$P = (16 \text{ tendons})(26.6) = 425.6 \text{ k}$$

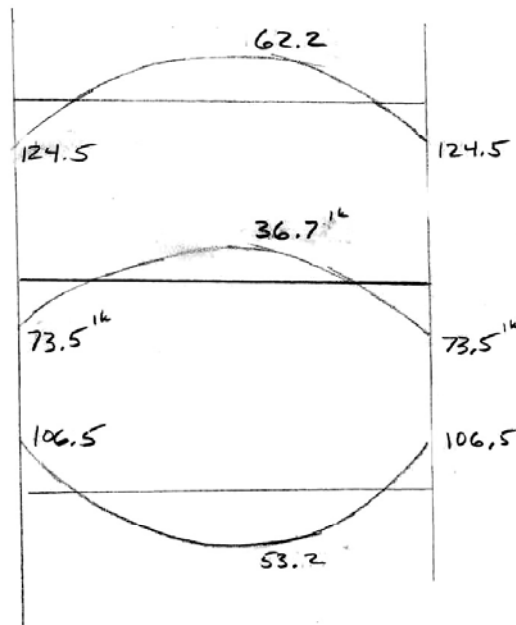
$$P/A = \frac{425.6 \text{ k}(1000)}{1260 \text{ in}^2} = 337.8 \text{ psi} > 125 \text{ psi ok}$$

$$w_a = \frac{(425.6)(8)(4.5''/12)}{30^2} = 1.42 \text{ klf}$$

$$w_{DL} = 1.66 \text{ klf}$$

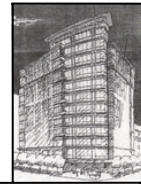
$$w_{LL} = .98 \text{ klf}$$

$$w_b = 1.42 \text{ klf}$$



2/4

# EXECUTIVE TOWER NW WASHINGTON, DC



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After jacking

mid span stress

$$f_{top} = \frac{(62.2 - 53.2)(12)(1000)}{1260} - 337.8 \text{ psi} = -252.1 < 164 \text{ psi ok}$$

$$f_{bot} = \frac{(-62.2 + 53.2)(12)(1000)}{1260} - 337.8 = -423.5 > -1800 \text{ psi ok}$$

Support

$$f_{top} = \frac{(-124.5 + 106.5)(12)(1000)}{1260} - 337.8 = -509.2 > -1800 \text{ psi ok}$$

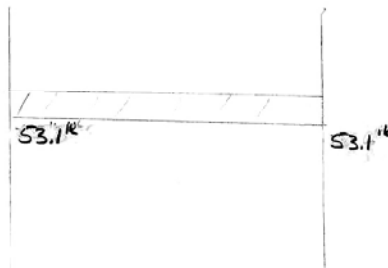
$$f_{bot} = \frac{(124.5 - 106.5)(12)(1000)}{1260} - 337.8 = -166.4 < 164 \text{ psi ok}$$

ultimate

$$M_1 = P \cdot e$$

$$= 425.6^k (4.5') / 12 = 159.6^k$$

$$M_{sec} = M_{bot} - M_1 = 106.5 - 159.6 = 53.1^k$$



Mid

$$M_u = 1.2(62.2) + 1.6(36.7) - 1.0(53.1) = 86.26^k$$

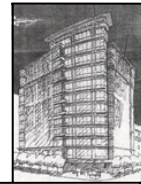
Support

$$M_u = 1.2(124.5) + 1.6(-73.5) + 1.0(106.5) = 160.5^k$$

3/4

# EXECUTIVE TOWER

## NW WASHINGTON, DC



SEAN HOWARD  
STRUCTURAL

### Reinforcement

		ft-k	ft	ft+4/ft	#4	check
8026	CS .60	48.2	8.7	5.54	11.8	ok
	MS .40	32.1	8.7	3.69	4.6	ok
-160.5	CS .75	120.4	8.7	13.84	11.8	2.04 (1)
	MS .25	40.1	8.7	4.61	4.6	ok

#4 @ 12" o.c.

C.S.

$$A_s = .20 \text{ in}^2/\text{ft}$$

$$d = 13.25 \text{''}$$

$$a = \frac{.20(60)}{.85(4)(12)} = .29 \text{''}$$

$$\phi M_n = \phi (.2)(60)(13.25 - \frac{.29}{2}) / 12 = 11.8 \text{ k/ft}$$

M.S.

$$A_s = .20 \text{ in}^2/\text{ft}$$

$$d = 5.25 \text{''}$$

$$a = .29 \text{''}$$

$$\phi M_n = \phi (.2)(60)(5.25 - \frac{.29}{2}) / 12 = 4.6 \text{ k/ft}$$

$$(1) M_n = 2.04(8.7) = 17.7 \text{ k}$$

try #6

$$a = \frac{.44(60)}{.85(4)(8.7 \times 12)} = .07 \text{''}$$

$$\phi M_n = \phi (.44)(60)(5.25 - \frac{.07}{2}) / 12 = 10.3 \text{ k} \text{ } \underline{\text{no good}}$$

USE 2 #6